

BIRLA INSTITUTE OF TECHNOLOGY, MESRA, RANCHI  
(END SEMESTER EXAMINATION)

CLASS: BTECH  
BRANCH: MECHANICAL

SEMESTER : V  
SESSION : MO/2022

SUBJECT: ME331 THERMO-FLUID ENGINEERING

TIME: 3:00 Hours

FULL MARKS: 50

INSTRUCTIONS:

1. The question paper contains 5 questions each of 10 marks and total 50 marks.
  2. Attempt all questions.
  3. The missing data, if any, may be assumed suitably.
  4. Before attempting the question paper, be sure that you have got the correct question paper.
  5. Tables/Data hand book/Graph paper etc. to be supplied to the candidates in the examination hall.
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Q.1(a) Define the final form of Reynold Transport Theorem and describe the significance of each term. [2]

Q.1(b) A two dimensional flow is described in the Lagrangian system as [4]

$$x = x_0 e^{-kt} + y_0 (1 - e^{-2kt}) \text{ and } y = y_0 e^{kt}$$

Find (i) the equation of path line of the particle and (ii) the velocity components in Eulerian system.

Q.1(c) A velocity field is given by  $\vec{V} = Ax\hat{i} - Ay\hat{j}$ ; x and y in meters.  $A = 0.3 \text{ s}^{-1}$ . [4]

Find out (i) equation of the streamlines in the xy plane and (ii) position at t=6s of particle located at (2, 8) at t=0

Q.2(a) Write the Bernoulli equation and its assumptions. [2]

Q.2(b) Consider a flow field represented by the stream function  $\psi = -\frac{A}{2\pi}(x^2 + y^2)$ , where A = [3]  
constant. Is this a possible two-dimensional incompressible flow? Is the flow irrotational?

Q.2(c) The x component of velocity in a steady, incompressible flow field in the xy plane is  $u = A(x^5 - 10x^3y^2 + 5xy^4)$ , where  $A = 2 \text{ m}^{-4} \cdot \text{s}^{-1}$  and x is measured in meters. [5]

(i) Find the simplest y component of velocity for this flow field.

(ii) Evaluate the acceleration of a fluid particle at point (x, y)=(1, 3).

Q.3(a) Specify the X, Y and Z momentum equations in a cartesian coordinate. [2]

Q.3(b) Consider a fully developed laminar flow through a straight tube of radius R of circular cross-section. [8]  
Consider the z-axis along the flow direction and the r-axis normal to it.

Find out the following with proper valid assumptions:

- i. continuity equation
- ii. velocity profile across the tube cross-section
- iii. shear stress profile across the tube cross-section
- iv. discharge through the pipe

Q.4(a) What is reversible work and irreversibility of a system? [2]

Q.4(b) A house that is losing heat at a rate of 80,000 kJ/h when the outside temperature drops to 15°C is to be heated by electric resistance heaters. If the house is to be maintained at 22°C at all times, determine the reversible work input for this process and the irreversibility. [3]

Q.4(c) Steam enters a turbine at 12 MPa, 550°C, and 60 m/s and leaves at 20 kPa and 130 m/s. The turbine is not adequately insulated and it estimated that heat is lost from the turbine at a rate of 150 kW. The power output of the turbine is 2.5 MW. Assuming the surroundings to be at 25°C, determine (a) the reversible power output of the turbine, (b) the exergy destroyed within the turbine, and (c) the second-law efficiency of the turbine. [5]

Given:  $h_1 = 3481.7 \text{ kJ/kg}$  and  $s_1 = 6.6554 \text{ kJ/kg K}$  at  $P_1 = 12 \text{ MPa}$ ,  $T_1 = 550^\circ \text{C}$

$h_2 = 2491.1 \text{ kJ/kg}$  and  $s_2 = 7.5535 \text{ kJ/kg K}$  at  $P_2 = 20 \text{ kPa}$

at dead state  $h_0 = 104.83 \text{ kJ/kg}$  at  $T_0 = 25^\circ \text{C}$

PTO

- Q.5(a) How and why is the stagnation enthalpy  $h_0$  defined? How does it differ from ordinary (static) enthalpy? [2]
- Q.5(b) What is the effect of back pressure on the development of normal shock in a convergent-divergent nozzle? [3]
- Q.5(c) Carbon dioxide enters an adiabatic nozzle at 1200 K with a velocity of 50 m/s and leaves at 400 K. [5]  
Assuming constant specific heats at room temperature, determine the Mach number ( $a$ ) at the inlet and ( $b$ ) at the exit of the nozzle. The gas constant of carbon dioxide is  $R = 0.1889$  kJ/kg·K. Its constant pressure specific heat and specific heat ratio at room temperature are  $c_p = 0.8439$  kJ/kg·K and  $k = 1.288$ .

::::28/11/2022::::M