BIRLA INSTITUTE OF TECHNOLOGY, MESRA, RANCHI (END SEMESTER EXAMINATION)

CLASS: BRANCH	M.TECH I: SER	SEMESTER : I SESSION : MO/18	
TIME:	SUBJECT: SR503 SPACE ENGINEERING AND SPACE DYNAMICS 3 HRS.	FULL MARKS: 50	
INSTRUC 1. The o 2. Atter 3. The r 4. Befor 5. Table	CTIONS: question paper contains 5 questions each of 10 marks and total 50 marks. npt all questions. missing data, if any, may be assumed suitably. re attempting the question paper, be sure that you have got the correct questi es/Data hand book/Graph paper etc. to be supplied to the candidates in the exa	on paper. amination hall.	
Q.1(a) Q.1(b)	Discuss in brief about the space and upper atmosphere environment. Write in detail about the advanced mission concepts.		[5] [5]
Q.2(a) Q.2(b)	Describe briefly about the factors causing dispersion of rockets in ground launch.A two-stage planetary exploration vehicle is launched from a high-orbit satellitevacuum trajectory. The following data are given:Flight and velocity increment in gravity-free vacuum6200 m/sEffective exhaust velocity, c (all stages)3050 m/sInitial launch vehicle mass4500 kgPropellant mass fraction (each stage)0.88Structural mass fraction (each stage)0.12Determine the payload for cases: (i) when the two stage masses are equal, and (i ratios of the two stages are equal.	into a gravity-free i) when the mass	[5] [5]
Q.3(a)	A satellite is launched into an elliptical orbit of Earth with an initial perigee altitude of 490 km and an apogee altitude of 1790 km. What are the velocities at these points? Given values for K=GM= 3.99×10^{14} m ³ /s ² , Earth's mean radius, R ₀ = 6.371×10^6 m. Derive and analyze the condition for impulsive shot.		[5]
Q.3(b)			[5]
Q.4(a) Q.4(b)	What is the basic purpose behind the analysis of spacecraft re-entry? Define a control analyzing the entry of a spacecraft into the Earth's atmosphere and obtain the base with the help of simplifying assumptions, obtain their dimensionless forms.	oordinate system for lance equations.	[5] [5]
Q.5(a)	Identify design drivers for a spacecraft intended for exploration of outer planets. I design based on the identified design drivers.	Present a conceptual	[10]

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